UNIT Video Switching Unit (VSU) OMG NO. 2294823-502, 504 SHUTTLE CCTV FMEA NO. _1.2.2 CHITICAL ITEMS LIST CHILICALITY _2/2_._ FAILURE EFFECT FATTURE MODE AND BATIONALE FOR ACCEPTANCE ON END ITEM CAUSE Na video autput No video signal to any DESIGN FEATURES destination including The VSU is a microprocessor-based video switching unit using an RCA 1802 on-board monitors and Causes: microprocessor, CNOS RAM, and TTL PRON. Computer 1/0. decoding logic, digital audio (1) Command failure on downlink. and switch control circuitry are implemented in CMOS CU4000 series logic to minimize microgrocessor, 1/0. power dissipation. The design incorporates DMOS FEI devices (SO211s) purchased to A1-2592389-501 or 2294889-502 Horst Case: an RCA spec control drawing (SCD) as the basic video switch element. Video (2) Timing logic, Loss of mission split-screen capability incorporates glass delay line modules procured from Logic board (VSU A3) critical video. Microsonics (originally Corning) to an RCA SCD. The video amplifier design uses 2592392-504 or 2294891-502 monolithic NE5539 wideband up amps in a fashion similar to the sync amp design (3) Switch matrix short on output employed in the RCU. Amplifier A6, A7, or A8, 2294894-502 Parts were required to be JAN reliability level parts of their equivalent. Part (4) Power bus short, all VSU boards selection falls into three categories: (1) JAN or better parts from the Military QPL. (2) Parts demonstrated to NASA to be equivalent to JAN level via test data (e.g., CD4000/3N series parts), or (3) Parts procured to an RCA spec control drawing which calls out tests and screening to effect JAN equivalency. BARE BOARD DESIGN [A6, A7, A8] The design of the associated A6, A7, and A8 boards is constructed from laminated copper-clad epoxy glass sheets (NENA G-10) Grade FR-4), PER MIL-P-55617A. Circuit connections are made through printed traces which run from point to point on the board surfaces. Every trace terminates at an annular ring. The annular ring surrounds the hole in which a component lead or terminal is located. This ring provides a footing for the solder, ensuring good mechanical and electrical performance. Its size and shape are governed by HIL-P-\$5640 as are trace widths, spacing and routing. These requirements are reiterated specifically in drawing notes to further assure compliance. Variations between the artwork master and the final product (due to irregularities of the etching process) are also controlled by drawing notes. This prevents making defective boards from good artwork, Holes which house no lead or terminal, but serve only to electrically interconnect the different

board layers, contain stitch bars for mechanical support and increased reliability.

The thru holes are drilled from a drill tape thus eliminating the possibility of human error and allowing light control over hole and annular ring concentricity, an important reliability criterion. After drilling and etching, all copper cladding is tin-lead plated per MIL-STD-1495. This provides for easy and reliable soldering at

the time of board assembly, even after periods of prolonged storage.

FHEA NO. 1.2.2 CRITICALITY 2/2	<u>·</u>	SHUTTLE CCTY CRETICAL LIEMS LIST	ONIT <u>Video Switching Unit (V</u> OHG NO. <u>2294823-502, 504</u> SHEET <u>2</u> OF <u>7</u>
	FAILURE EFFECT ON END. ITEM No video signal to any destination including an-board monitors and downlink. Norst Case: Loss of mission critical video.	RATIONALE FOR ACCEPT. DESIGN FEATURES BOARD ASSEMBLY DESIGN (A6, A7, A8) All components are installed in a manner which assure Component leads are pre-tinned, allowing total wetting are formed to provide stress relief and the badies of Special mounting and handling instructions are includater final assembly. The board is coated with ureablumidity and contamination. BUARD PLACEMENT (A1, A3, A6, A7, A8) The (A1, A3, A6, A7, and A8) boards are secured in the by gold-plated beryllium copper card guides. Connewith blind-mated connectors. Disengagement during spans the board's free edge. BARE BOARD CONSTRUCTION (A1, A3) The boards are of "welded wire" construction. At the distinguish it from a normal PC board except that he generally are not connected to PC traces. Only the ground potentials to the ICs are on PCs. An annular board where each power and ground pin is located. The trace like any other component lead. Aside from a construction techniques used in PC board layout at BOARD ASSEMBLY (A1, A3) The drilled and etched boards are populated with several pins. Power and ground pins, as well as concluded. Discreet temponents (resistors, diodes, capabifurcated terminals, where they are smidered. Flad by-lead, to the tops of the weld pins. After well-by-lead, to the tops of the weld pins. After well-by-lead, to the tops of the weld pins.	SHEET 2 OF
		trimmed away. Circuit connections are made using #2 wire is welded to the pin surfaces on the board back using a machine which is tape driven, thus eliminated due to operator error. All wiring & circuit perform hox-level installation. After successful testing, they drawing notes and the assembly is coated with ursuit the board is inserted in the box on card-edge guides PC boards.	O ANG nickel weld wire. The side. All wire welds are done ing the possibility of miswiring wante is tested prior to components are staked as required there.

FNEA NO. 1.2.2		SHUTTLE CCTV CRITTEAL ITEMS LIST	UNIT <u>Video Switching Unit (V</u> DWG NO. <u>2294823-502. 504</u> SHEET <u>3</u> OF 7	
FAILURE HODE AND FAILURE EFFECT		RATIONALE FOR ACCEPTANCE		
CAUSE o video output ouses: 1) Command failure on microprocessor, 1/0, A1-2592389-501 or 2294889-502 2) Liming logic, Logic board (VSU A3) 2592392-504 or 2294891-502 3) Switch matrix short on output Amplifier A6, A7, or A8, 2294894 502 4) Power bus short, all VSU boards	No video signal to any destination including on-board monitors and downlink. Worst Case: Loss of mission critical video.	QUALIFICATION TEST For Qualification Test Flow, see Table 2 located ACCEPTANCE TEST The CCTV systems' VSU is subjected to the follow Vibration: 20-80Hz: 3 dB/Oct-rise fr 80-350 Hz: 0.04 G ² /Hz 350-750 Hz: 3 dB/Oct-Fall to 750-1000: 0.018 G ² /Hz 1000-2000: 3 dB/Oct-Fall to	ATION TEST iffication Test Flow, see Table 2 located at the front of this book. KE TEST systems' VSU is subjected to the following testing: ation: 20-80Hz: 3 dB/Oct-rise from 0.01 6 ² /Hz to 0.04 G ² /Hz 80-350 Hz: 0.04 G ² /Hz 350-750 Hz: 3 dB/Oct-Fall to 0.018 G ² /Hz 750-1000: 0.018 G ² /Hz 1000-2000: 3 dB/Oct-Fall to 0.009 G ² /Hz Test Duration: 1 Minute per Axis	
2338		0° f: Time to stablize equipm 100° f: Time to stablize equipm 100° f: Time to stablize equipm for Acceptance Test Flow, see Table 1 located at the VSU may not have been subjected to the vacuu OPERATIONAL TEST In order to verify that CCTV components are open health of all the command related components fro	0° f: Time to stablize equipment plus 1 hour 100° f: 11me to stablize equipment plus 1 hour ceptance Test flow, see Table 1 located at the front of this book. I may not have been subjected to the vacuum condition. 10NAL TEST er to verify that CCTV components are operational, a test must verify the of all the command related components from the PHS (A7A1) panel switch.	
		through the REU, through the sync lines to the Camera/PIU, to the Camera/PIU command decoder. The test must also verify the camera's ability to produce video, the VSU's ability to route video, and the monitor's ability to display video. A similar test would be performed to verify the HOM command path. Pre-launch on Orbiter Test/In-flight lest 1. Power CCTV System. 2. Via the PHS panel, select a monitor as destination and the camera under test as source. 3. Send "Camera Power On" command from PHS panel. 4. Select "External Sync" on monitor. 5. Observe video displayed on monitor. Note that if video on monitor is synchronized (i.e., stable raster) then this indicates that the camera is receiving composite sync from the ROU and that the camera is producing synchronized video. 6. Send Pan, Filt, Focus, Zoum, ALC, AND Gamma commands and visually (either via the monitor or direct observation) verify operation. 7. Select downlink as destination and camera under test as source. 8. Observe video routed to downlink.		
		 Send "Camera Power Off" command via PRS 10. Repeat Steps 3 through 9 except issue of This proves that the CCIV equipment is 	commands via the MDM command path.	

FREA NO1.2.2		SHUTTLE CCTV CRITICAL ITEMS LIST	ONET <u>Video Switching Unit (VS</u> OWG NO. <u>2294823-502, 504</u> SHEET <u>4</u> OF <u>7</u>
fallure More and CAUSE To video output Sauses: 1) Command failure on	FASLURE EFFECT ON FND 11EM No video signal to any destination including an-board monitors and downlink. Norst Case: Loss of mission critical video.	QA/INSPECTION Procurement Control — The YSU Parts and hardware items vendors and suppliers, which meet the requirements set Quality Plan Work Statement (WS-2593176). Resident Dispracurement documents to establish the need for GSC on Incoming Inspection and Storage — Incoming Quality improved materials and parts. Results are recorded by drawing and control numbers for future reference and subjected to incoming acceptance tests as called for Test Instructions. Incoming flight parts are further RCA 1846684 — Preconditioning and Acceptance Requirement the exception the OPA and PIND testing is not performing spected per PAI 316 — Incoming Inspection Instructions Processing Incoming or Purchased Parts Designated for are delivered to Haterial Controlled Stores and retain until fabrication is required. Non-conforming material Board (HRB) disposition. (PAI-307, PAI IQC-531). Buard Assembly A Test — Prior to the start of VSU board assembly adjainst the as-built-parts-list (ABPL). DCAS are designated for all printed circuit, wire wrap and harness connectors for suddering wiring, crimping, so workmanship prior to coating of the component side of Specific VSU board assembly and test instructions are applicable documents are called out in the Fabrication (FN-2294023) and parts list PL 2294823. These incluid Process Standard RIV-566 2280881, Process Standard - I Specification - Ordering 2280749, Specification Name P Specification - Urethane coating 270077, Specification Specification - Workmanship 8030035, Specification - Bond Specification - Workmanship 8030035, Specification - Bond Specification - Workmanship 8030035, Specification - Specification - Specific	s are procured from approved to forth in the CCTV contract and CAS personnel review all in selected parts (PAI 517). spections are made on all ylot and retained in file by traceability. All EEE parts are in PAI 315 - Incoming Inspection processed in accordance with ents for Electronic Parts, with ents for Mechanical items are one for mechanical items. PAI s, and PAI 612 - Procedure for flight Use. Accepted items and under specified conditions als are held for Material Review of assembly, all items are extensional items are extensional accomplated to form who assembly all items are extensional series and conditions wilded wire boards, plus lider splices and quality boards and sleeving of harnesses, provided in drawing notes, and in Procedure and Record de wire connection List 2295906, Bonding Velcro Tape 2280889, late Application 1960167, ding and Staking 2280878, on - locking compound 2026116, Marking 228087b.
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FMEA NO. 1.2.2		SHUTTLE COTY CHITICAL ITEMS LIST	UNIT Video Switching Unit (VS DWG NO. 2294823-502 504 7
Couses: (1) Command failure on downline of microprocessor, 1/0, A1-2592369-501 or 2294869-502 toss	FAILURE EFFECT ON END TITH No video signal to any destination including on-board monitors and downlink. Worst Case: Loss of anission critical video.	QA/INSPECTION (Continued) YSU Assembly and Jest An open box test is performed per FP-IT-22944832, and TP-AT-2294823, including vibration and thermal vacuum witnessed, traceability numbers are recorded and call to use. RCA quality and DCAS inspections are performs specified FPR operations in accordance with PAI-204, DCAS personnel witness VSU button-up and critical temporation acceptance tasts and review test data/results after all repair, rework and retest. Preparation for Shipment - The VSU is packaged according a second according to the packaging and handling guidelines. All assembly drawing, parts list, ABPL, test data, etc. thousand the folder assigned specifically to each as retained for reference. An EIDP is prepared for each requirements of MS-2593176. RCA QC and DCAS personning and marking, and review the EIDP for complete.	d an Acceptance Test per a. Torques are specified and ibrated tools are chack prior ned at the completion of PAI-205, PAI-206, and PAI-217. rquing. RCA and BCAS personnels. These personnel also inspect ding to 2200746. Process related documentation including is gathered and held in a ssembly. This folder is h VSU in accordance with the el witness crating, packaging,
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SHATTLE CCTV CRITICAL ITEMS LIST TAILURE FREET ON LING ITEM No video output Causer (1) Command failure on eferoprotestar. 1/0 214-259/28-501 or 229489-502 (2) List charact (VSLA) 25/29/29-504 or 229489-502 (3) Suitch matrix short on output Angliffer An, A. T. or An, 229489-502 (4) Yomer bus shart, all YSU boards The command of the c				
Causes: (1) Command failure on wicroprocessor. 170. A1-2592389-501 or 2294889-502 (2) Timing lugic. Lugic board (VSU A3) 2592392-504 or 2294891-502 (3) Switch matrix short on output Amplifier A6, A7, or A8, 2294894-502 (4) Power bus short. all VSU boards ON END ITEM No video signal to any destination including on-board monitors and downlink. Horst Case: Loss of mission critical video. Description: Acceptance Test Failure, Board Level, Ambient Environment. Unit power & ground pins were found shorted together. Bench test, buard assembly, ambient temperature. Cause: These shorts were caused by the A1 board PNMP traces, design error (1st item testing). Corrective Action: ECN was written to correct PNMP & net list. Board reworked per per ECN #CCT-586. All A1 boards built prior to or subsequent to this failure were repaired, or				
	CAUSE No video output Causes: (1) Command failure on	ON END ITEM No video signal to any destination including on-board monitors and downlink. Horst Case: Loss of mission	FAILURE HISTORY TOR HOSTI-Log 0348 VSU - 2294823-501 S/N F001 Description: Acceptance Test Failure, Board Lev ground pins were found shorted together. Ben temperature. Cause: These shorts were caused by the Al board f (1st item testing). Corrective Action: ECN was written to correct PW per ECN #CCT-586. All Al boards built prior to or subsequent to this	rel, Ambiest Environment. Unit power & such test, buard assembly, ambiest PNMP traces, design error TP & net list. Board reworked per

FREA NO. 1.2.2		SHUITLE CCTV CRITICAL ITEMS LIST	OWIT <u>Video Switching Unit (VSU</u> OWG NO. <u>2294873-502, 504</u> SMEET <u>7</u> OF <u>7</u>	
FAILURE MODE AND FAILURE EFFECT CAUSE ON END ITEM		RATIONALE FOR ACCEPTANCE		
CAUSE. Causes: (1) Command failure na microprocessor, 1/0, A1-2592389-501 or 2294889-502 (2) Timing logic, Logic board (VSB A3) 2592392-504 or 2294891-502 Switch matrix short on output Amplifier A6, A7, or A8, 2294894-502 [4) Power bus short, all VSD boards	No video signal to any destination including on-board monitors and downlink. Worst Case: Loss of mission critical video.	Loss of video. Possible loss of major mission objectives due to loss of RMS cameras or other required cameras. CREH ACTIONS If possible, continue RMS operations using alternative visual cues. CREH TRAINING Crew should be trained to use possible alternatives to CCTV. MISSION CONSTRAINT Where possible, procedures should be designed so they can be accomplished without CCTV.		